

14 CFR 25.103, Stall Speed

Part 25 AIRWORTHINESS STANDARDS: TRANSPORT CATEGORY AIRPLANES
Subpart B--Flight Performance: Turbine Engine Powered Airplanes

[(a) The reference stall speed, V_{SR} , is a calibrated airspeed defined by the applicant. V_{SR} may not be less than a 1-g stall speed. V_{SR} is expressed as:

$$V_{SR} \geq \frac{V_{CLMAX}}{\sqrt{n_{ZW}}}$$

where: V_{CLMAX} = Calibrated airspeed obtained when the load factor- corrected coefficient

$$\left(\frac{n_{ZW} W}{qS} \right)$$

is first a maximum during the maneuver prescribed in paragraph (c) of this section. In addition, when the maneuver is limited by a device that abruptly pushes the airplane at a selected angle of attack (e.g., a stick pusher), V_{CLMAX} may not be less than the value existing at the instant the device operates;

n_{ZW} = Load factor normal to the flight path at V_{CLMAX}

W = Airplane gross weight;

S = Aerodynamic reference wing area; and

q = Dynamic pressure.

(b) V_{CLMAX} is determined with:

- (1) Engines idling, or, if that resultant thrust causes an appreciable decrease in stall speed, not more than zero thrust at the stall speed;
- (2) Propeller pitch controls (if applicable) in the takeoff position;

- (3) The airplane in other respects (such as flaps and landing gear) in the condition existing in the test or performance standard in which V_{SR} is being used;
 - (4) The weight used when V_{SR} is being used as a factor to determine compliance with a required performance standard;
 - (5) The center of gravity position that results in the highest value of reference stall speed; and
 - (6) The airplane trimmed for straight flight at a speed selected by the applicant, but not less than $1.13V_{SR}$ and not greater than $1.3V_{SR}$.
- (c) Starting from the stabilized trim condition, apply the longitudinal control to decelerate the airplane so that the speed reduction does not exceed one knot per second.
- (d) In addition to the requirements of paragraph (a) of this section, when a device that abruptly pushes the nose down at a selected angle of attack (e.g., a stick pusher) is installed, the reference stall speed, V_{SR} , may not be less than 2 knots or 2 percent, whichever is greater, above the speed at which the device operates.]

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Subpart B--Flight Stalls