

Airworthiness Directive

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

Amendment 39-3085; AD **76-21-04**

GENERAL ELECTRIC Models CF6-6D and CF6-6D1 Engines
PDF Copy (If Available):

▼ Preamble Information

AGENCY: Federal Aviation Administration, DOT

DATES: Effective November 30, 1977.

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76-21-04 GENERAL ELECTRIC: Amendment 39-2747 as amended by Amendment 39-3085. Applies to Models CF6-6D and CF6-6D1 Turbofan Engines.

Compliance required by June 30, 1978, unless previously accomplished.

To prevent excessive overpressure in the high pressure compressor, remove the abradable material from the inside diameter of the Fan Stator Shroud Mid Ring (Booster Stage) in accordance with General Electric Service Bulletin (CF6-6) 72-647 or subsequent FAA Approved Revision thereto.

The manufacturer's service bulletins identified and described in this directive are incorporated herein and made a part hereof pursuant to 5 U.S.C. 552(a)(1). All persons affected by this directive who have not already received these documents from the manufacturer may obtain copies upon request to General Electric Company, Cincinnati, Ohio 45215. These documents may also be examined at the FAA Great Lakes Region, 2300 E. Devon Avenue, Des Plaines, Illinois 60018 and at FAA headquarters, 800 Independence Avenue, S.W., Washington, D.C. A historical file on this AD which includes the incorporated material in full is maintained by the FAA at its headquarters in Washington, D.C., and at the Great Lakes Region.

Amendment 39-2747 became effective October 20, 1976.

This amendment 39-3085 becomes effective November 30, 1977.